

SUPPLEMENTAL TYPE CERTIFICATE

10056001

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

ABLE AEROSPACE, Inc.

**7706 EAST VELOCITY WAY
MESA CA 85212
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.R.512

Type Certificate Holder: BELL HELICOPTER TEXTRON CANADA

Type: BELL 206/407

Model: 206A, 206B, 206L, 206L-1,
206L-3, 206L-4

Description of Design Change:

EASA Approval of Able Aerospace PMA Strap Assembly P/N 2AA-310-004-101/103 eligible for installation on Bell 206 rotorcraft as per PMA No. PQ2487NM Supplement No.17 dated May 07, 2014.

Note: This STC is issued in accordance with § 2.2.3 point (b) of the Technical Implementation Procedure (Rev. 5) of EU/USA Bilateral Agreement.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 16 December 2015



Massimo MAZZOLETTI
Head of Rotorcraft Department

10035781
SUPPLEMENTAL TYPE CERTIFICATE - 10056001 - ABLE AEROSPACE, Inc. - 307772



EASA Certification Basis:

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

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Associated Technical Documentation:

Supplemental Instruction for Continued Airworthiness Report 5421-AL, Rev. NC, Dated January 27, 2014.

or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)" and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Certification Compliance Report CCR-5421, Rev. NC Dated April 22, 2014.

Drawing AA-5421-000 Rev. NC

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -